

# On the Stiffness Matrix of the Intervertebral Joint: Application to Total Disc Replacement

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## Abstract

The traditional method of establishing the stiffness matrix associated with an intervertebral joint is valid only for infinitesimal rotations, whereas the rotations featured in spinal motion are often finite. In the present paper, a new formulation of this stiffness matrix is presented which is valid for finite rotations. This formulation uses Euler angles to parameterize the rotation, an associated basis, which is known as the dual Euler basis, to describe the moments, and it enables a characterization of the non-conservative nature of the joint caused by energy loss in the poroviscoelastic disc and ligamentous support structure. As an application of the formulation, the stiffness matrix of a motion segment is experimentally determined for the case of an intact intervertebral disc and compared to the matrices associated with the same segment after the insertion of a total disc replacement system. In this manner, the matrix is used to quantify the changes in the intervertebral kinetics associated with total disc replacements. As a result, this paper presents the first such characterization of the kinetics of a total disc replacement.

## Index Terms

Spine kinematics, intervertebral disc, stiffness matrix, disc arthroplasty.

## 1 INTRODUCTION

While there are hopes of seeing vertebral disc replacement travel the same successful path as total hip and knee replacements, the complexity of the joint structure between pairs of vertebrae has caused unforeseen complications.<sup>1</sup> The intervertebral disc has a complex structure and function that includes synergistic functioning with the facets in constraining motion and supporting load. These structural complexities obscure optimal design choices since the relative motion of vertebra is non-trivial to characterize and measure. More importantly, inappropriate modifications to this motion may lead to other problems such as osteoarthritis in the facet joints and motion segment instability, which may lead to impingement of neural structures [5]. Spine mechanics are further complicated by a loading regime that consists of bending moments and loads that are multi-directional and often coupled.

A wide-range of measurements are currently being used to characterize spinal movements within the orthopaedic research community, including: range of motion (see, e.g., [6]), disc pressure (see, e.g., [7]), neutral zone [8], helical axis of motion (see, e.g., [9], [10]), vertebral strain (see, e.g., [11]), facet forces (see, e.g., [12], [13]), and stiffness (see, e.g., [14]). Collectively

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1. For further background on total disc replacements, see [1]–[4] and references therein.

22 this has provided a vast amount of information on the motion of the spine. Much of this data is  
 23 crucial in the design and development of effective total disc replacements (TDR). Of particular  
 24 interest in this paper is an examination of the stiffness changes induced by a TDR.

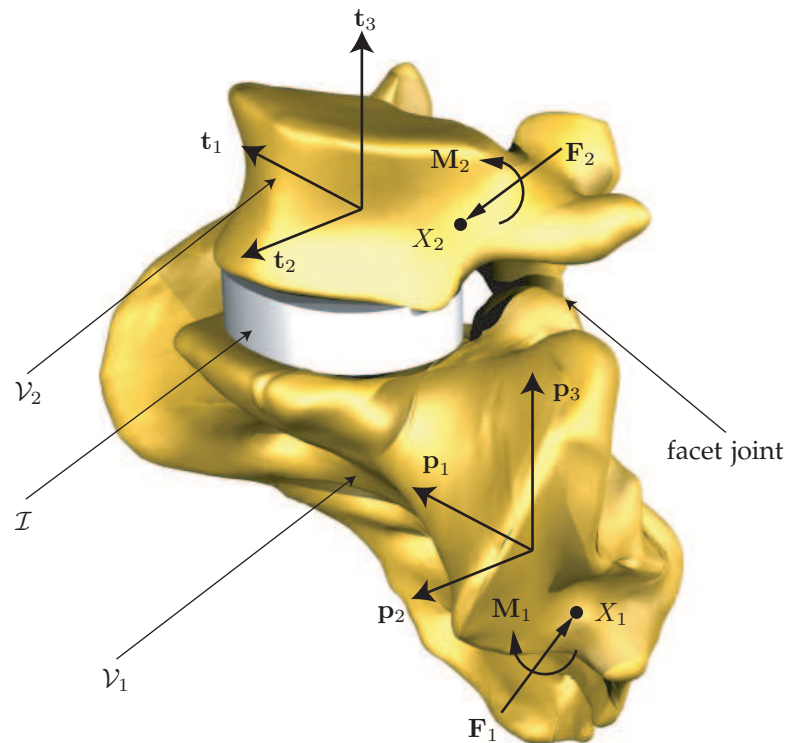


Fig. 1. Schematic of a motion segment consisting of a pair of vertebral bodies  $\mathcal{V}_1$  and  $\mathcal{V}_2$  and the intervertebral disc  $\mathcal{I}$ . One of the pair of facet joints is also indicated, as are the bases  $\{\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3\}$  for  $\mathcal{V}_1$  and  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$  for  $\mathcal{V}_2$ . For the image shown in this figure, the lower body is S1 and the upper body is L5.

25 To examine the stiffness changes induced by a TDR, one is first lead to the seminal paper  
 26 by Panjabi et al. [14] which was published in 1976. In [14], a stiffness matrix characterizing  
 27 a six degree-of-freedom vertebral motion segment in the thoracic spine was proposed. Using  
 28 symmetry arguments, restricting attention to infinitesimal rotations, and assuming certain sym-  
 29 metries, the number of stiffnesses in this matrix were reduced from 36 to 12. Subsequent work  
 30 by Gardner-Morse, Stokes et al. [15]–[18], have measured these 12 parameters. A related stiffness  
 31 matrix for the lumbar spine was proposed by McGill and Norman [19], and in subsequent works  
 32 the potential energy of the muscle forces and external forces was incorporated into this matrix  
 33 (see Cholewicki and Norman [20], Howarth et al. [21], McGill and Bennett [22], and references  
 34 cited therein).

35 Unfortunately, the stiffness matrices proposed by Panjabi et al. [14] and McGill and Norman  
 36 [19] have several restrictions which limit their utility. The most problematic is the inability  
 37 to accommodate finite rotations and energy losses due to the poroviscoelastic nature of the  
 38 intervertebral disc and the nonconservative forces and moments due to the facet joints and

39 ligaments. These and other deficiencies are addressed in this paper by presenting an alternative  
40 method of calculating the stiffness matrix of a motion segment. The segment in question consists  
41 of two vertebral bodies, their adjoining intervertebral disc, the facet joints, and the ligaments  
42 connecting the two bodies. The construction of the stiffness matrix is performed with the help of  
43 the developments in O'Reilly [23] and O'Reilly and Srinivasa [24], and by exploiting a recently  
44 developed basis which is known as the dual Euler basis. The methodology is valid for finite  
45 rotations and can accommodate the (non-conservative) forces and moments due to the facet  
46 joints and ligaments. Thus, the matrix proposed in this paper will be non-symmetric due to the  
47 nonconservative forces that are included in the model.

48 The use of the dual Euler basis in the present paper is similar to the use of a related dual  
49 basis in Howard et al. [25] and Žefran and Kumar [26] which recently came to the attention  
50 of the authors. In this pair of papers, Žefran et al. use screw theory to describe the wrench  
51 (force and moment) components with respect to a dual basis and use these components to  
52 establish a stiffness-twist relationship. Their basis couples the individual components of the  
53 twists (displacements and rotations), and their work could also be used to formulate a stiffness  
54 matrix for the motion segment.<sup>2</sup>

55 The primary aim of the present paper is to introduce the theory which supports this new  
56 formulation of the stiffness matrix. Secondly, a method for distilling the 36-component matrix  
57 into a single scalar for statistical purposes is presented. The third and final aim is to demonstrate  
58 the value of both the stiffness matrix and its respective scalar by applying them experimentally  
59 to characterize the kinetics of a TDR. In particular, these metrics are used to evaluate the sagittal  
60 placement of the SYNTHES PRODISC-L TDR system and compare it to an intact vertebral disc.  
61 The results presented in this paper are the first such characterization of a TDR system.

62 An outline of the paper is as follows. In the following section, the parameterization of the  
63 displacement and relative rotation of a pair of vertebra is discussed. In the interests of conciseness,  
64 many of the details on the parameterization are placed in Appendix A. Section 2 contains  
65 a presentation of the stiffness matrix of a motion segment and a discussion of several of its  
66 unusual features. Most of the details on the derivation of this stiffness matrix are presented in  
67 Appendix B. A discussion of the kinetics of a motion segment follows. The application of the  
68 stiffness matrix  $K$  to the characterization of a TDR forms the primary focus of Section 3. In  
69 particular, the experimental measurements of  $K$  for an intact disc and three distinct placements  
70 of a TDR are presented. The paper closes with discussions of the objectives of the paper and how  
71 they were achieved, and the directions of future research on  $K$ . For the readers' convenience a  
72 section on nomenclature follows Section 4.

## 73 2 THEORY

74 A motion segment consists of two vertebral discs, an intervertebral disc, a pair of facet joints  
75 and the muscles and ligaments connecting the vertebra (cf. Fig. 1). The relative motion of the  
76 discs can be characterized by a set of three displacements and three Euler angles. The stiffness  
77 matrix  $K$  relates a set of forces and moments to the three displacements and three angles.

### 78 2.1 Kinematics

79 The three-dimensional displacement vector  $y$  is defined by the relative motion of two points  $X_1$   
80 and  $X_2$ , one on each vertebra. Although the selection of these points is arbitrary, their selection  
81 will effect the stiffness matrix. To define the Euler angles a pair of right-handed orthonormal  
82 bases is needed. One of these basis, which is denoted by  $\{p_1, p_2, p_3\}$  is fixed to the lower vertebra,

2. For further details on the necessities of using dual bases to describe moments and wrenches, the reader is referred to [23], [26], [27].

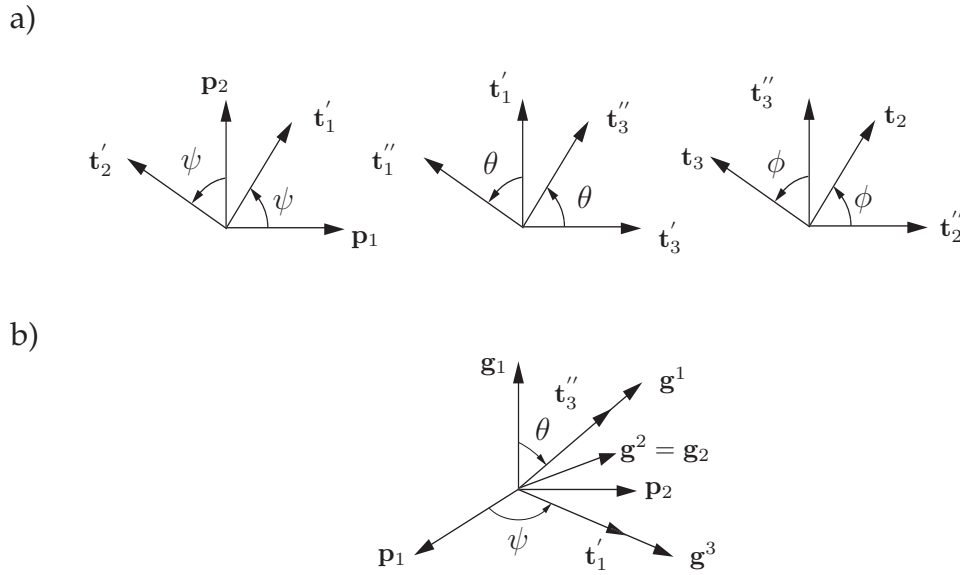


Fig. 2. Schematic of the 3-2-1 set of Euler angles:  $\psi$ ,  $\theta$ , and  $\phi$ . In this figure, the vectors  $\mathbf{g}_1 = \mathbf{p}_3$ ,  $\mathbf{g}_2 = \mathbf{t}'_2 = \cos(\psi)\mathbf{p}_2 - \sin(\psi)\mathbf{p}_1$ , and  $\mathbf{g}_3 = \mathbf{t}''_1 = \cos(\theta)\mathbf{t}'_1 + \sin(\theta)\mathbf{p}_3$  form the Euler basis. As illustrated in b), the dual Euler basis  $\{\mathbf{g}^1, \mathbf{g}^2, \mathbf{g}^3\}$  is distinct from the Euler basis.

83 and the other, which is denoted by  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$  is affixed to the upper vertebra. An example  
 84 featuring the L5/S1 motion segment is shown in Fig. 1.

In studies on the kinematics of the spine, it is standard to refer to the angles as axial rotation ( $\psi$ ), lateral bending ( $\theta$ ), and flexion-extension ( $\phi$ ). Referring to Fig. 2(a), the axial rotation represents a rotation about  $\mathbf{p}_3$  through an angle  $\psi$ . This is followed by a lateral bending about  $\mathbf{t}'_2 = \cos(\psi)\mathbf{p}_1 + \sin(\psi)\mathbf{p}_2$ . The final angle of rotation is a flexion-extension  $\phi$  about  $\mathbf{t}_1$ . The axes of rotation  $\mathbf{p}_3, \mathbf{t}'_2$ , and  $\mathbf{t}_1$  define the Euler basis:

$$\begin{bmatrix} \mathbf{g}_1 \\ \mathbf{g}_2 \\ \mathbf{g}_3 \end{bmatrix} = \begin{bmatrix} \mathbf{p}_3 \\ \mathbf{t}'_2 \\ \mathbf{t}_1 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 \\ -\sin(\psi) & \cos(\psi) & 0 \\ \cos(\theta)\cos(\psi) & \cos(\theta)\sin(\psi) & -\sin(\theta) \end{bmatrix} \begin{bmatrix} \mathbf{p}_1 \\ \mathbf{p}_2 \\ \mathbf{p}_3 \end{bmatrix}. \quad (1)$$

85 Based on the choice of axes, the set of Euler angles used here is known as the 3-2-1 set, and,  
 86 as discussed by Crawford et al. [28], this is the optimal choice of Euler angles for the motion  
 87 segment. Further details on the Euler angles used in this paper, the Euler basis and the dual  
 88 Euler basis can be found in Appendix A.

## 89 2.2 The Stiffness Matrix K

To define the stiffness matrix  $\mathbf{K}$ , one presumes that one can measure the resultant force and moment on one of the vertebra. For the upper vertebra, the resultant force is denoted by  $\mathbf{F}_2$  and the resultant moment, relative to  $X_2$ , is denoted by  $\mathbf{M}_2$ . Correspondingly, the resultant force on the lower vertebra is denoted by  $\mathbf{F}_1$  and the resultant moment, relative to  $X_1$ , is denoted by  $\mathbf{M}_1$ . When one measures these forces and moments and then correlates them to the displacements  $\mathbf{y}$  and relative rotations  $\psi, \theta$ , and  $\phi$ , the forces and moments when the displacements and relative rotations are zero will not necessarily vanish. These residual forces and moments are denoted by a subscript 0. The stiffness matrix is then defined by the relationship

$$\mathbf{F} = \mathbf{F}_0 - \mathbf{K}\mathbf{d}. \quad (2)$$

In Eq. (2), the generalized force vector  $F$ , the generalized residual force vector  $F_0$ , the generalized displacement vector  $d$ , and stiffness matrix  $K$  are

$$F = \begin{bmatrix} \mathbf{F}_2 \cdot \mathbf{p}_1 \\ \mathbf{F}_2 \cdot \mathbf{p}_2 \\ \mathbf{F}_2 \cdot \mathbf{p}_3 \\ \mathbf{M}_2 \cdot \mathbf{g}_1 \\ \mathbf{M}_2 \cdot \mathbf{g}_2 \\ \mathbf{M}_2 \cdot \mathbf{g}_3 \end{bmatrix}, \quad F_0 = \begin{bmatrix} \mathbf{F}_{20} \cdot \mathbf{p}_1 \\ \mathbf{F}_{20} \cdot \mathbf{p}_2 \\ \mathbf{F}_{20} \cdot \mathbf{p}_3 \\ \mathbf{M}_{20} \cdot \mathbf{g}_1 \\ \mathbf{M}_{20} \cdot \mathbf{g}_2 \\ \mathbf{M}_{20} \cdot \mathbf{g}_3 \end{bmatrix}, \quad d = \begin{bmatrix} \mathbf{y} \cdot \mathbf{p}_1 \\ \mathbf{y} \cdot \mathbf{p}_2 \\ \mathbf{y} \cdot \mathbf{p}_3 \\ \psi \\ \theta \\ \phi \end{bmatrix}, \quad K = \begin{bmatrix} k_{11} & \cdots & k_{16} \\ \vdots & \ddots & \vdots \\ k_{61} & \cdots & k_{66} \end{bmatrix}. \quad (3)$$

90 The residual force  $F_{20}$  and residual moment  $M_{20}$  are the respective values of  $F_2$  and  $M_2$  when  
91 the displacement  $d = 0$ .

There are several unusual features in Eq. (2). First, as shown in Eqs. (22) and (26) of Appendix B, the forces  $F_1$  and  $F_2$  are equal and opposite, as are the moments  $M_1$  and  $M_2$ :

$$\mathbf{F}_1 = -\mathbf{F}_2, \quad \mathbf{M}_1 = -\mathbf{M}_2. \quad (4)$$

92 Second, it is necessary to compute the components  $M_2 \cdot \mathbf{g}_k$ , and as the Euler basis vectors  $\mathbf{g}_k$   
93 depend on the Euler angles  $\theta$  and  $\psi$  these components are often not intuitive. Indeed, as discussed  
94 in the Appendix, computing  $M_2 \cdot \mathbf{g}_k$  is equivalent to expressing  $M_2$  in terms of its components  
95 relative to the dual Euler basis  $\{\mathbf{g}^1, \mathbf{g}^2, \mathbf{g}^3\}$ .<sup>3</sup>

96 In comparison to the stiffness matrix presented by Panjabi et al. [14], it is unrealistic to expect  
97 that  $K$  will be symmetric.<sup>4</sup> However, if attention is restricted to infinitesimal rotations, and  
98 the symmetry restrictions of Panjabi et al. are imposed, then  $K$  will simplify to the stiffness  
99 matrix proposed in [14].<sup>5</sup> The moment components determined by Panjabi et al.'s stiffness matrix  
100 correspond to  $M \cdot \mathbf{p}_k$ . Unfortunately, it has long been known [23] that a constant moment  $M_0 \mathbf{p}_3$ ,  
101 where  $M_0$  is a constant, is nonconservative when finite rotations are present. However, the  
102 moment  $M_0 \mathbf{g}^1$  is conservative. The use of the components  $M_2 \cdot \mathbf{g}_k$  in (3) are precisely to ensure  
103 that when  $K$  is symmetric, then  $F_2 - F_{20}$  and  $M_2 - M_{20}$  are guaranteed to be conservative even  
104 in the presence of finite rotations.

### 105 2.3 An Aggregate Stiffness Ratio

Comparison of the stiffness matrices for two motion segments on a term by term basis is difficult  
and often not very illuminating. An alternative strategy, which is proposed here, is to define a  
aggregate stiffness to be the norm of the stiffness matrix:

$$k = \sqrt{\text{tr}(KK^T)}, \quad (5)$$

where  $\text{tr}$  denotes the trace of a matrix. To compare the aggregate stiffness of two motion  
segments, one can then define a normalized value  $S$ :

$$S = \frac{k_{\text{I}} - k_{\text{II}}}{k_{\text{I}}}, \quad (6)$$

106 where  $k_{\text{I}}$  and  $k_{\text{II}}$  are the aggregate stiffnesses associated with the respective stiffness matrices  
107 of the two motion segments. The aggregate stiffness ratio  $S$  is distinct from the stability indices  
108 discussed in Howarth et al. [21]. Indeed, as one cannot expect the stiffness matrices to be  
109 symmetric or positive definite, such stability indices may not be revealing.

3. Details on the transformation of components of vectors relative to the bases used in this paper are summarized in Appendix C.

4. It is well-known in structural dynamics that the presence of nonconservative forces and moments can destroy the symmetry of the stiffness matrix.

5. The precise details on this equivalence can be found in Appendix B.1



### 110 3 STIFFNESS ALTERATIONS DUE TO TOTAL DISK REPLACEMENTS

111 To demonstrate the utility of the stiffness matrix presented in this paper, the present section  
 112 details its application to a data set that has recently been collected to determine the sensitivity  
 113 of TDR placement along the sagittal plane.

#### 114 3.1 Experimental Protocol

##### 115 Specimen Preparation

116 Healthy, non-degenerate fresh-frozen L5/S1 motion segments were harvested from human spines  
 117 ( $n=5$ , mean age: 44, three females and two males). Specimen preparation included meticulous  
 118 removal of muscular tissue so as to retain the integrity of the capsular and ligamentous elements.  
 119 Afterwards, the specimens were potted in polymethylmetacrylate (PMMA), so that the S1 end-  
 120 plate was parallel to the PMMA surface and clamping faces.

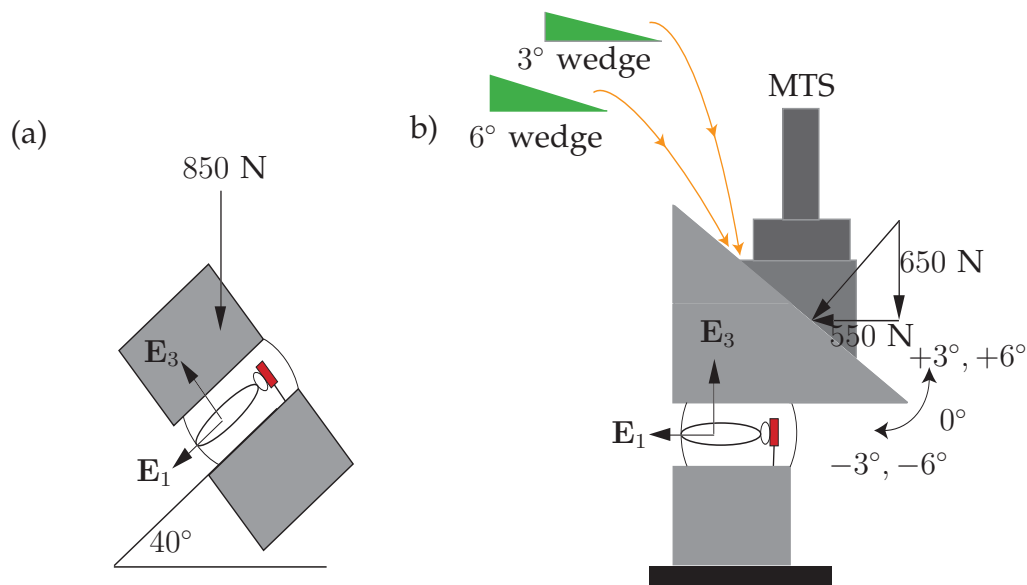


Fig. 3. Schematic diagram of experimental set-up: 40° sacral slope and 850 N load in standing position: (a), Testing device constrained L5 posture in flexion, extension, and bending for investigating L5/S1 kinematics and (b), load is uniformly distributed and applies both shear of 550 N and compression of 650 N. In (b), the 3° and 6° wedges which are used to achieve the desired relative motion of the vertebrae are also shown.

##### 121 Mechanical Testing

122 Each specimen was placed in a servo-hydraulic apparatus (Bionix 858, MTS Systems Corp. Eden  
 123 Meadow, MN) such that the disc was oriented at 40° relative to the horizontal axis (Fig. 3)  
 124 as described previously in Rousseau et al. [12], [29]. The specimens were loaded with 850 N  
 125 generating 650 N of disc compression and 550 N of horizontal shear consistent with free body  
 126 analyses of L5/S1 based on specific morphometric studies.<sup>6</sup> Wedges were added at the frictionless  
 127 interface to impose 3° and 6° of flexion, extension, and lateral bending postures, while axial  
 128 torsion was unconstrained. The 12° total range of motion in the sagittal and the frontal plane  
 129 was below the normal physiological zone of the L5/S1 joint [5].

130 Once tested with the disc intact, a TDR was performed (ProDisc-L, Synthes Inc. West Chester,  
 131 PA USA). This particular device has a polyethylene (UHMWPE) on metal (CoCrMo) bearing

6. See, [19], [30]–[33].

132 interface with a non-retentive ball-and-socket design allowing 3 degrees-of-freedom. The device  
 133 was initially positioned 3 mm ( $\pm 0.5$  mm) posterior to the center of the inferior (S1) endplate.  
 134 The specimen was tested in this position, and the device was then moved forward 3 mm to  
 135 the central location and tested, followed by 3 mm anterior. This enabled measurement of the  
 136 sensitivity of device placement along the sagittal plane.

137 Specimen preconditioning consisted of three cycles of complete loading and unloading prior  
 138 to testing in each posture and was reduced to one cycle when the specimen was instrumented.  
 139 During testing, data were collected after one minute of loading for each posture. Tissues were  
 140 kept moist during testing by wrapping in saline-soaked gauze. A three-camera optoelectronic  
 141 system (Motion Analysis Corpl, Santa Rosa, CA) was used to track the motion between the two  
 142 vertebral bodies, while a load cell rigidly attached to S1 simultaneously recorded the resultant  
 143 force and moments.

### 144 3.2 Data Analysis

145 Kinematic data was computed using software that integrated data from the load cell and motion  
 146 analysis files. An optimization algorithm was applied to the optical targets between neutral (no  
 147 wedge) and each of the rotated postures to get the optimal estimates of the Euler angles and the  
 148 translation of a marker on L5 for each motion.<sup>7</sup> In this manner, the six components necessary  
 149 to resolve the displacement vector  $d$  for each motion were determined.

150 Load cell data for each motion was translated into the six components of the generalized force  
 151 vector  $F$ . The difference between rotated postures and the neutral posture were determined to  
 152 give relative forces and moments. These vectors were then transformed to the dual Euler basis  
 153 as described in the two earlier sections to allow accurate calculations despite the relatively  
 154 large angles of rotation. The result is a six component vector  $F - F_0$  consisting of three forces  
 155 components and three moment components.

The displacement and load vectors for the rotations were organized into 6-by-6 matrices,  $d_E$   
 and  $F_E$  respectively, where each column represents a different motion, indicated by  $m_1, \dots, m_6$ :

$$d_E = [ \{d\}_{m_1} \quad \dots \quad \{d\}_{m_6} ], \quad F_E = [ \{F - F_0\}_{m_1} \quad \dots \quad \{F - F_0\}_{m_6} ]. \quad (7)$$

156 To compute the stiffness matrix  $K$ , one then computes  $F_E (d_E)^{-1}$ . The resulting matrix is com-  
 157 posed of 36 stiffness coefficients relative to a neutral posture.

To analyze the TDRs influence on the stiffness matrix  $K$  of the motion segment, the stiffness  
 matrix of the intact disc is compared to the corresponding matrix of the motion segment after the  
 implantation of the TDR. The former and latter matrices are labeled by  $K_I$ , and  $K_T$ , respectively.  
 Following Eq. (6), the stiffness ratio  $S$  was then computed:

$$S = \frac{k_T - k_I}{k_I}, \quad (8)$$

158 where  $k_{T,I}$  are the aggregate stiffnesses associated with the matrices  $K_{T,I}$ .

### 159 3.3 Results

160 The four resultant stiffness matrices (intact and three device placements) varied considerably  
 161 from specimen to specimen. In the interests of brevity, the outcomes are demonstrated with  
 162 the stiffness matrices for a single representative specimen, and the aggregate stiffness ratio  $S$   
 163 relied upon for interpreting general changes between device positions among the five motion  
 164 segments.

7. The algorithm is based on the TRIAD algorithm and a classical optimal estimate of the translation. Discussions of these  
 optimal estimates can be found in several papers, e.g., Dorst [34], Shuster and Oh [35], Spoor et al. [10], [36], and Woltring et  
 al. [37].

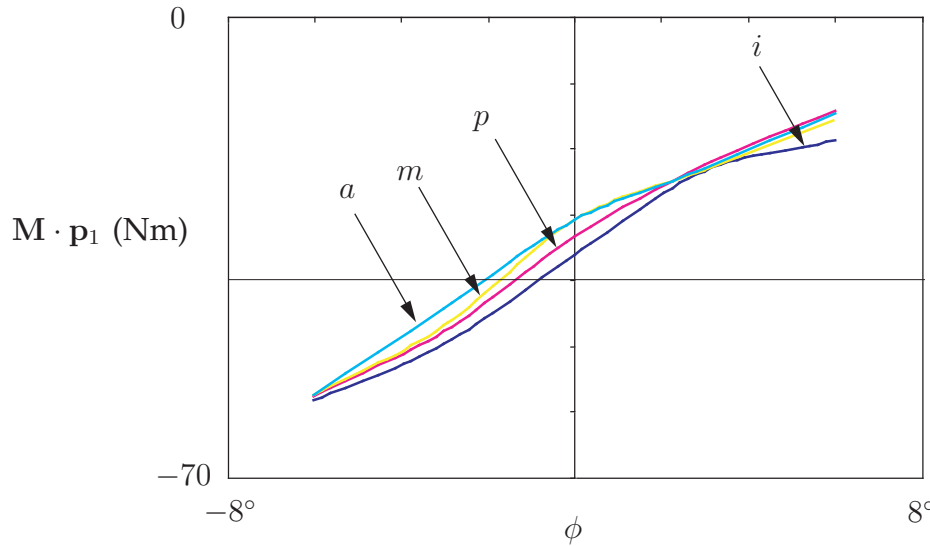


Fig. 4. The moment component  $M \cdot p_1$  as a function of the angle  $\phi$  of flexion/extension for an intact motion segment and a three different positionings of a TDR. Here, and in Figs. 5 and 6, the label  $i$  stands for intact,  $p$  stands for posterior,  $a$  denotes anterior, and  $m$  denotes a centered positioning of the TDR.

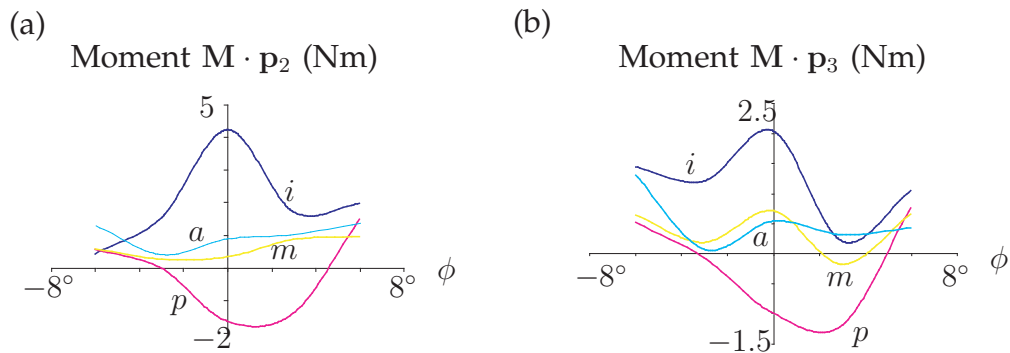


Fig. 5. The moment components (a)  $M \cdot p_2$  in the lateral direction and (b)  $M \cdot p_3$  in the axial direction as a function of the angle  $\phi$  of flexion/extension for an intact motion segment and a three different positionings of a TDR. The label  $i$  stands for intact,  $p$  stands for posterior,  $a$  denotes anterior, and  $m$  denotes a centered positioning of the TDR.

165 **3.3.1 Stiffness Matrices**

For one of the five specimens, the following stiffness matrices were computed. The first of these matrices,  $K_i$  is for the intact specimens, while the matrices  $K_{p,m,a}$  correspond to the respective posterior, middle and anterior placements of the TDR:

$$K_i = \begin{bmatrix} 80307 & 155477 & 29398 & -65.43 & -1102 & 1892.3 \\ 34377 & 382690 & 91836 & -113.8 & -1779 & 1585.5 \\ 8882.4 & -17381 & 3227.2 & -400 & -537.3 & 1630.6 \\ 599.5 & 1439.9 & 399.83 & -11.52 & -11.97 & 13.468 \\ 7544.9 & 16460 & 2832 & -50.85 & 174.02 & -276.9 \\ -105 & 7014.3 & 1468.5 & 7.9949 & -154.8 & 307.05 \end{bmatrix},$$



$$\begin{aligned}
K_p &= \begin{bmatrix} 40731 & 31188 & 9704.2 & -315.8 & 35.422 & -393.2 \\ 59442 & 248555 & 52209 & -1510 & 1225 & -752.3 \\ 24364 & -46694 & 11257 & -977.4 & -1116 & 4754.9 \\ -74.46 & 560.93 & 66.933 & -1.935 & 8.0895 & -38.99 \\ -552.3 & -1369 & -53.01 & 107.87 & -81.85 & -454.9 \\ -5468 & -3523 & -973 & 97.05 & -23.56 & -234.9 \end{bmatrix}, \\
K_m &= \begin{bmatrix} 36784 & 1954.5 & -8646 & 672.07 & 689.97 & -1651 \\ 39612 & 127890 & 62070 & -2068 & 4910.3 & -8295 \\ 4175.6 & 15301 & -6291 & 922.22 & -1258 & 2181.2 \\ 162.18 & 120.06 & 196.37 & -12.96 & 1.6026 & -12.87 \\ 6924.9 & 4106.4 & -623.5 & 107.96 & 49.332 & -147.8 \\ -2413 & 3323.1 & 547.01 & 28.409 & -153.3 & 227.34 \end{bmatrix}, \\
K_a &= \begin{bmatrix} 18016 & 18719 & -8076 & 800.31 & -164.2 & -506 \\ -39459 & 186498 & 23518 & 1019.8 & -60.67 & -2705 \\ 22129 & -9158 & 1813.8 & -312.5 & 235.24 & 767.71 \\ 218.16 & 63.657 & 124.48 & -8.551 & -3.891 & 1.0397 \\ 4050.3 & 2576.9 & -1038 & 86.583 & -20.16 & -23.57 \\ -461.4 & -2808 & 574.83 & -70.38 & -8.937 & 103.03 \end{bmatrix}. \tag{9}
\end{aligned}$$

166 It is interesting to note that some of the diagonal stiffness elements are negative. In further  
167 contrast to the stiffness matrices reported in the literature, the matrices presented above are not  
168 symmetric. Concerning units, the displacements and rotations used to measure these matrices  
169 had units of meters and radians, respectively. Likewise, the forces and moments were computed  
170 using unit of Newtons and Newton meters. As a result, the stiffnesses have distinct units, for  
171 example,  $K_{11}$  has units of Newtons/meter,  $K_{16}$  and  $K_{61}$  have units of Newtons, and  $K_{45}$  has  
172 units of Newton meters.

### 173 3.3.2 Residual Forces and Moments and Negative Stiffnesses

174 The experimental set-up resulted in substantial moments of extension calculated about the center  
175 of the intact disc in the neutral position (see Fig. 4). This moment was calculated using the  
176 identity  $\mathbf{M} = \mathbf{M}_m + \boldsymbol{\pi} \times \mathbf{F}_m$  where  $\boldsymbol{\pi}$  is the position vector of the center of mass of the  
177 intact disc relative to the load cell and  $\mathbf{F}_m$  and  $\mathbf{M}_m$  are the force and moment measurements  
178 from the load cell. Furthermore, torsional and lateral bending moments were present during  
179 flexion and extension (see Fig. 5). Since the stiffness coefficients are calculated from the force  
180 and displacement vectors relative to the neutral posture, it is useful to display these results. It  
181 should be noticed from these figures that residual values of the moment  $\mathbf{M}$  are present even  
182 when the angle  $\phi = 0^\circ$ , and the slopes of these graphs are consistent with some of the negative  
183 values for individual stiffnesses that were found. In many cases, the motion segment had a  
184 more rigid response in the neutral posture than in the rotated postures, especially once the  
185 device was inserted. This can be explained by the high elastic modulus of the device which  
186 resists axial loads, but low coefficient of friction between the UHMWPE and the chrome-moly  
187 upon bending.

188 For each specimen, the aggregate stiffness ratio,  $S$ , between the instrumented and intact motion  
189 segment was calculated for the three device positions. The value of  $S$  is dimensionless and can  
190 be thought of as a fractional change from the intact disc. For instance, if the value of  $S$  is 0.5, the  
191 device caused the motion segment to respond 50% more rigidly than the intact disc on average  
192 over the six degrees-of-freedom. It was found that the average  $S$  of the five specimens that were  
193 tested was not significantly different ( $P \leq 0.5$ ) for any of the device positions (Fig. 6), but there  
194 was a trend of increasing stiffness as the device was moved posteriorly.

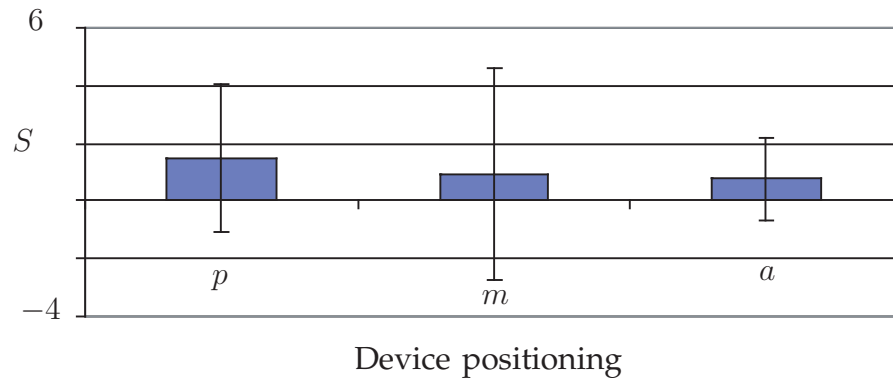


Fig. 6. The values of the aggregate stiffness ratio  $S$  for various positionings of a TDR.

## 4 DISCUSSION

195

196 The objective of this paper was twofold. First, a new method of calculating the stiffness matrix  
 197 that provides accurate calculations for large, more physiologic angles of rotation was described.  
 198 This method is one of the only two possible formulations of this matrix which is valid for  
 199 finite rotations.<sup>8</sup> The principal difference in the computation of the stiffness matrix for the finite  
 200 rotation case compared to the classical counterpart is the need to use the values of the angles  
 201  $\theta$ ,  $\phi$ , and  $\psi$ , when computing the moment components.

202

203 As a second objective, the stiffness matrix was used to compare the changes in kinetics induced  
 204 by a TDR. Although several research efforts aimed at characterizing the kinematical changes  
 205 induced by a TDR have appeared (see, e.g., [12], [29]), this paper has presented the the first kinetic  
 206 comparison of a TDR to an intact disc. Three placements of the TDR were also considered in  
 207 this comparison. In an attempt to distill the tremendous amount of data into a possibly clinically  
 208 relevant metric, an aggregate stiffness ratio  $S$  was introduced. This ratio compares two matrices  
 209 (in this case, instrumented versus intact) and distills the result into a single metric. Preliminary  
 210 results of the  $S$  ratio calculations (cf. Fig. 6) display its ability as an efficient tool to compare and  
 211 contrast devices as it is easy to interpret both clinically and statistically. There is a clear need  
 212 for such a metric, apparent by the stream of technology that has poured into the orthopaedic  
 213 spine community in the past decade. While further work is needed to prove its efficacy, it has  
 214 the potential for quantifying device stiffness in vitro.

214

215 The stiffness matrix introduced in this paper is unique from those presented in earlier works  
 216 since, as described above, the kinematic values are measured relative to a neutral or pre-loaded  
 217 state. Additionally, previous studies in spinal kinetics have typically calculated each stiffness  
 218 coefficient independently or assumed their value from symmetry. By combining six displacement  
 219 vectors and their respective force vectors the stiffness matrix presented in this paper represents  
 220 a comprehensive stiffness measure in all six degrees-of-freedom. Present work by the authors  
 221 involves performing an extensive error analysis which aims to quantify how accurately one can  
 222 determine the stiffness matrix  $K$  given the limitations inherent in the measurements of rotations,  
 displacements, forces, and moments.

8. The second formulation, which would, in principle, feature screw theory and be based on the developments of Howard et al. [25] and Žefran and Kumar [26], remains to be fully developed.

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## 229 NOMENCLATURE

- $\{\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3\}$  = right-handed orthonormal basis  
 $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$  = right-handed orthonormal basis  
 $\{\mathbf{t}'_1, \mathbf{t}'_2, \mathbf{t}'_3\}$  = right-handed orthonormal basis  
 $\{\mathbf{t}''_1, \mathbf{t}''_2, \mathbf{t}''_3\}$  = right-handed orthonormal basis  
 $\{\mathbf{g}_1, \mathbf{g}_2, \mathbf{g}_3\}$  = Euler basis  
 $\{\mathbf{g}^1, \mathbf{g}^2, \mathbf{g}^3\}$  = dual Euler basis  
 $\psi, \theta, \phi$  = angles of axial rotation, lateral bending, and flexion-extension, respectively  
 $\boldsymbol{\omega}, \boldsymbol{\omega}_1, \boldsymbol{\omega}_2$  = angular velocity vectors  
 $\mathbf{x}, \mathbf{x}_1, \mathbf{x}_2$  = position vectors  
 $\mathbf{y} = \mathbf{x}_2 - \mathbf{x}_1$  = displacement vector  
 $\mathbf{R}$  = rotation matrix  
 $\mathbf{K}$  = stiffness matrix of motion segment  
 $\mathbf{K}_u$  = stiffness matrix owing to conservative contributions  
 $U$  = potential energy function  
 $\mathbf{F}_{C_1}, \mathbf{F}_{C_2}$  = conservative forces  
 $\mathbf{M}_{C_1}, \mathbf{M}_{C_2}$  = conservative moments  
 $\mathbf{F}_m$  = force measured by load cell  
 $\mathbf{M}_m$  = moment measured by load cell  
 $\mathbf{F}_{V_1}, \mathbf{F}_{V_2}$  = viscoelastic forces  
 $\mathbf{M}_{V_1}, \mathbf{M}_{V_2}$  = viscoelastic moments  
 $\mathbf{F}_{nc_1}, \mathbf{F}_{nc_2}$  = nonconservative forces  
 $\mathbf{M}_{nc_1}, \mathbf{M}_{nc_2}$  = nonconservative moments  
 $\mathbf{F}, \mathbf{F}_0$  = generalized force vectors  
 $\mathbf{d}$  = generalized displacement vector  
 $k$  = norm of the stiffness matrix  
 $S$  = Aggregate stiffness ratio

## 230 APPENDIX A

### 231 BACKGROUND ON ROTATIONS, EULER ANGLES AND THE DUAL EULER BASIS

232 The rotation of interest is the relative rotation of a pair of vertebra  $\mathcal{V}_1$  and  $\mathcal{V}_2$ . To parameterize  
 233 the transformation induced by this rotation is a set of Euler angles is used. In this Appendix,

234 relevant background on the Euler angles is presented which is based on the authoritative review  
 235 by Shuster [38] and supplemented by material on the dual Euler basis from [23], [27].

236 In what follows, it is presumed that a set of right-handed orthonormal basis vectors  $\{\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3\}$   
 237 are affixed to  $\mathcal{V}_1$  and a similar set  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$  are attached to  $\mathcal{V}_2$  (see Fig. 1 and Fig. 7). The rotation  
 238 of interest can be considered as transforming  $\mathbf{p}_1$  to  $\mathbf{t}_1$ ,  $\mathbf{p}_2$  to  $\mathbf{t}_2$ , and  $\mathbf{p}_3$  to  $\mathbf{t}_3$ .

239 As may be seen from Fig. 2(a), the manner in which the Euler angles parametrize the rotation  
 240 is easily visualized by imagining two intermediate bases  $\{\mathbf{t}'_1, \mathbf{t}'_2, \mathbf{t}'_3\}$ ,  $\{\mathbf{t}''_1, \mathbf{t}''_2, \mathbf{t}''_3\}$ . The first angle  
 241  $\psi$  represents the rotation of  $\mathbf{p}_1$  and  $\mathbf{p}_2$  about  $\mathbf{p}_3$  to their respective transformed values  $\mathbf{t}'_1$  and  $\mathbf{t}'_2$ .  
 242 Similarly, the second rotation through the angle  $\theta$  about the vector  $\mathbf{t}'_2$  and it transforms  $\mathbf{t}'_3$  and  
 243  $\mathbf{t}'_1$  into  $\mathbf{t}''_3$  and  $\mathbf{t}''_1$ , respectively. The third rotation is through the angle  $\phi$  about the vector  $\mathbf{t}''_1$ . This  
 244 final rotation transforms  $\mathbf{t}''_2$  and  $\mathbf{t}''_3$  into  $\mathbf{t}_2$  and  $\mathbf{t}_3$ , respectively.

One can define a proper-orthogonal matrix  $R$  to represent the transformation of  $\mathbf{p}_i$  to  $\mathbf{t}_i$ :

$$\begin{bmatrix} \mathbf{t}_1 \\ \mathbf{t}_2 \\ \mathbf{t}_3 \end{bmatrix} = \begin{bmatrix} R_{11} & R_{21} & R_{31} \\ R_{12} & R_{22} & R_{32} \\ R_{13} & R_{23} & R_{33} \end{bmatrix} \begin{bmatrix} \mathbf{p}_1 \\ \mathbf{p}_2 \\ \mathbf{p}_3 \end{bmatrix}, \quad \begin{bmatrix} \mathbf{p}_1 \\ \mathbf{p}_2 \\ \mathbf{p}_3 \end{bmatrix} = \begin{bmatrix} R_{11} & R_{12} & R_{13} \\ R_{21} & R_{22} & R_{23} \\ R_{31} & R_{32} & R_{33} \end{bmatrix} \begin{bmatrix} \mathbf{t}_1 \\ \mathbf{t}_2 \\ \mathbf{t}_3 \end{bmatrix}, \quad (10)$$

where the components of the matrix are

$$\begin{bmatrix} R_{11} & R_{12} & R_{13} \\ R_{21} & R_{22} & R_{23} \\ R_{31} & R_{32} & R_{33} \end{bmatrix} = \begin{bmatrix} c(\psi) & -s(\psi) & 0 \\ s(\psi) & c(\psi) & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} c(\theta) & 0 & s(\theta) \\ 0 & 1 & 0 \\ -s(\theta) & 0 & c(\theta) \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & c(\phi) & -s(\phi) \\ 0 & s(\phi) & c(\phi) \end{bmatrix}. \quad (11)$$

245 Here, the abbreviations  $c(x)$  for  $\cos(x)$  and  $s(x)$  for  $\sin(x)$  have been used. The three axes of  
 246 rotation for the individual angles associated with the set of Euler angles are known as the Euler  
 247 basis vectors. These unit vectors are denoted by  $\{\mathbf{g}_1, \mathbf{g}_2, \mathbf{g}_3\}$ . For the 3-2-1 set of Euler angles,  
 248 Eq. (1) provides a definition of  $\{\mathbf{g}_1, \mathbf{g}_2, \mathbf{g}_3\}$  in terms of the basis vectors  $\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3$ . Alternatively,  
 249 with the help of Eq. (10) and Eq. (11), one can express the Euler basis vectors in terms of the  
 250 basis vectors  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$ .

251 As can be verified from Eq. (1), the Euler basis vectors form a basis provided  $\theta \neq \pm\frac{\pi}{2}$ .  
 252 As a result, one restricts the second angle  $\theta \in (-\frac{\pi}{2}, \frac{\pi}{2})$  to ensure that the Euler basis vectors  
 253 form a basis. The angle  $\theta$  measures lateral bending and so this restriction is trivially satisfied  
 254 physiologically. The other two angles are free to range from 0 to  $2\pi$ .

The angular velocity vector  $\boldsymbol{\omega}$  associated with the rotation has a convenient representation  
 when the Euler basis vectors are used:

$$\boldsymbol{\omega} = \dot{\psi}\mathbf{p}_3 + \dot{\theta}\mathbf{t}'_2 + \dot{\phi}\mathbf{t}_1. \quad (12)$$

In the sequel a set of vectors are need which can extract from  $\boldsymbol{\omega}$  the angular speeds  $\dot{\psi}$ ,  $\dot{\theta}$ , and  $\dot{\phi}$ .  
 This set of vectors is known as the dual Euler basis vectors:  $\{\mathbf{g}^1, \mathbf{g}^2, \mathbf{g}^3\}$ .<sup>9</sup> By definition, the dual  
 Euler basis vectors satisfy the relations

$$\boldsymbol{\omega} \cdot \mathbf{g}^1 = \dot{\psi}, \quad \boldsymbol{\omega} \cdot \mathbf{g}^2 = \dot{\theta}, \quad \boldsymbol{\omega} \cdot \mathbf{g}^3 = \dot{\phi}. \quad (13)$$

That is,

$$\begin{aligned} \mathbf{g}_1 \cdot \mathbf{g}^1 &= 1, & \mathbf{g}_1 \cdot \mathbf{g}^2 &= 0, & \mathbf{g}_1 \cdot \mathbf{g}^3 &= 0, \\ \mathbf{g}_2 \cdot \mathbf{g}^1 &= 0, & \mathbf{g}_2 \cdot \mathbf{g}^2 &= 1, & \mathbf{g}_2 \cdot \mathbf{g}^3 &= 0, \\ \mathbf{g}_3 \cdot \mathbf{g}^1 &= 0, & \mathbf{g}_3 \cdot \mathbf{g}^2 &= 0, & \mathbf{g}_3 \cdot \mathbf{g}^3 &= 1. \end{aligned} \quad (14)$$

9. A thorough discussion of these basis vectors can be found in [23], [27].

Given the Euler basis vectors, one can use the nine equations Eq. (14) to compute expressions for the dual Euler basis vectors. After a series of straightforward manipulations, one would find that the dual Euler basis vectors have the representations

$$\begin{bmatrix} \mathbf{g}^1 \\ \mathbf{g}^2 \\ \mathbf{g}^3 \end{bmatrix} = \begin{bmatrix} \cos(\psi) \tan(\theta) & \sin(\psi) \tan(\theta) & 1 \\ -\sin(\psi) & \cos(\psi) & 0 \\ \cos(\psi) \sec(\theta) & \sin(\psi) \sec(\theta) & 0 \end{bmatrix} \begin{bmatrix} \mathbf{p}_1 \\ \mathbf{p}_2 \\ \mathbf{p}_3 \end{bmatrix}. \quad (15)$$

255 It is important to note that the vectors  $\mathbf{g}^1$  and  $\mathbf{g}^3$  do not have unit magnitude (cf. Fig. 2(b)).  
 256 Expressions for the dual Euler basis vectors in terms of  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$  can established using Eq. (11)  
 257 and Eq. (15)

If the Euler angles are infinitesimal, then, from Eq. (15), it is easy to see that

$$\mathbf{g}^1 \approx \mathbf{p}_3 \approx \mathbf{t}_3, \quad \mathbf{g}^2 \approx \mathbf{p}_2 \approx \mathbf{t}_2, \quad \mathbf{g}^3 \approx \mathbf{p}_1 \approx \mathbf{t}_1. \quad (16)$$

258 Related results hold for the Euler basis vectors  $\mathbf{g}_k$ . For the spinal applications of interest, the  
 259 angles of rotation are not infinitesimal and so the approximations Eq. (16) cannot be used.

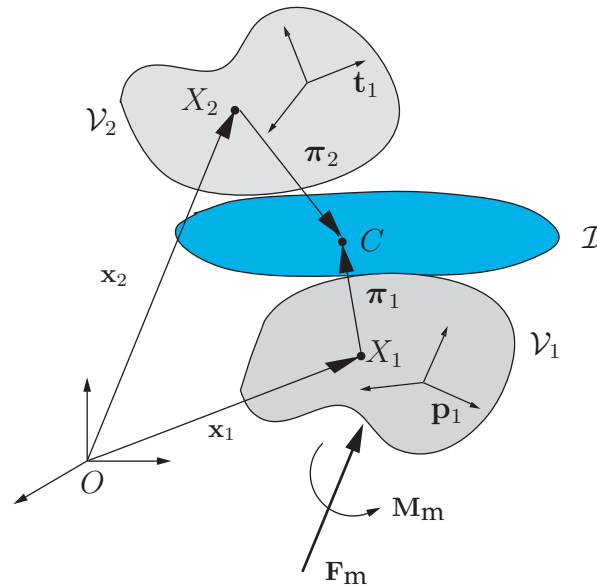


Fig. 7. A schematic drawing of an intervertebral disc  $\mathcal{I}$  between the vertebra  $\mathcal{V}_1$  and  $\mathcal{V}_2$ . The point  $C$  of the disc  $\mathcal{I}$  has a position vector  $\mathbf{x}_1 + \boldsymbol{\pi}_1 = \mathbf{x}_2 + \boldsymbol{\pi}_2$  relative to the fixed origin  $O$ . The force  $\mathbf{F}_m$  and moment  $M_m$  shown in this figure are supplied by the load cell.  
 fig:vertunit

## 260 APPENDIX B

### 261 DERIVATION OF THE STIFFNESS MATRIX OF A MOTION SEGMENT

262 Consider the system consisting of two vertebra  $\mathcal{V}_1$  and  $\mathcal{V}_2$  located on either side of an interverte-  
 263 bral disc  $\mathcal{I}$  shown in Fig. 7. Of interest in this paper is the development of a mechanical model  
 264 for the intervertebral disc and the facet joints. It is assumed that the disc and joints result in  
 265 a force  $\mathbf{F}_1$  and a moment  $\mathbf{M}_1$  on  $\mathcal{V}_1$  and a force  $\mathbf{F}_2$  and a moment  $\mathbf{M}_2$  on  $\mathcal{V}_2$ . The force  $\mathbf{F}_1$  is  
 266 assumed to act at the material point  $X_1$  of  $\mathcal{V}_1$  and the moment  $\mathbf{M}_1$  is taken relative to this point  
 267 (cf. Fig. 1). Similarly,  $\mathbf{F}_2$  is assumed to act at the material point  $X_2$  of  $\mathcal{V}_2$  and the moment  $\mathbf{M}_2$  is  
 268 relative to  $X_2$ . In an experimental apparatus to examine the kinetics of a segment of the spine, it  
 269 is standard to place a load cell directly under the vertebral body  $\mathcal{V}_1$ . The load cell provides two

270 sets of measurements: the three force components and three moment components:  $\mathbf{F}_{\mathbf{m}_k} = \mathbf{F}_{\mathbf{m}} \cdot \mathbf{p}_k$   
 271 and  $\mathbf{M}_{\mathbf{m}_k} = \mathbf{M}_{\mathbf{m}} \cdot \mathbf{p}_k$  where  $k = 1, 2, 3$ .

The rotation of  $\mathcal{V}_2$  relative to  $\mathcal{V}_1$  can be characterized by a rotation  $R$ . The rotation is parameterized in this paper using a set of a set of 3-2-1 Euler angles:  $\psi$ ,  $\theta$ , and  $\phi$ . Hence, the difference between the angular velocity vectors  $\boldsymbol{\omega}_1$  and  $\boldsymbol{\omega}_2$  of  $\mathcal{V}_1$  and  $\mathcal{V}_2$  can be expressed as<sup>10</sup>

$$\boldsymbol{\omega}_2 - \boldsymbol{\omega}_1 = \dot{\psi}\mathbf{g}_1 + \dot{\theta}\mathbf{g}_2 + \dot{\phi}\mathbf{g}_3. \quad (17)$$

The position vectors of the points  $X_1$  and  $X_2$  of the vertebrae are denoted by  $\mathbf{x}_1$  and  $\mathbf{x}_2$ , respectively. It is standard to express these vectors in terms of the fixed basis  $\{\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3\}$ , e.g.,  $\bar{\mathbf{x}}_1 = \sum_{k=1}^3 x_{1,k} \mathbf{p}_k$ . Furthermore, it is necessary to define the relative displacement vector of the point  $X_2$  relative to  $X_1$ :

$$\mathbf{y} = y_1\mathbf{p}_1 + y_2\mathbf{p}_2 + y_3\mathbf{p}_3 = \mathbf{x}_2 - \mathbf{x}_1. \quad (18)$$

272 Although, it is customary to choose  $X_1$  to be the center of mass of  $\mathcal{V}_1$  and  $X_2$  to be the center  
 273 of mass of  $\mathcal{V}_2$ , this choice is often not convenient. Further, precise identification of the center of  
 274 mass of a vertebra is non-trivial.

## 275 B.1 Potential Energy, Conservative Forces, and Conservative Moments

To postulate a potential energy for the motion segment and correlate its derivatives to the forces and moments on the vertebra, the methodology used in O'Reilly and Srinivasa [24] is followed.<sup>11</sup> The crucial assumption is that the potential energy for the conservative forces and conservative moments supplied by the facets, ligaments, and intervertebral disc is

$$U = U(\mathbf{y}, \psi, \theta, \phi). \quad (19)$$

In this case, the relative translation and rotation of the vertebra are independent. The forces ( $\mathbf{F}_{C_1}$  and  $\mathbf{F}_{C_2}$ ) and moments ( $\mathbf{M}_{C_1}$  and  $\mathbf{M}_{C_2}$ ) supplied by the disc, facets, and ligaments to the vertebrae are conservative:<sup>12</sup>

$$-\dot{U} = \mathbf{F}_{C_1} \cdot \dot{\mathbf{x}}_1 + \mathbf{F}_{C_2} \cdot \dot{\mathbf{x}}_2 + \mathbf{M}_{C_1} \cdot \boldsymbol{\omega}_1 + \mathbf{M}_{C_2} \cdot \boldsymbol{\omega}_2. \quad (20)$$

As

$$\dot{U} = \sum_{i=1}^3 \frac{\partial U}{\partial y_i} \dot{y}_i + \frac{\partial U}{\partial \psi} \dot{\psi} + \frac{\partial U}{\partial \theta} \dot{\theta} + \frac{\partial U}{\partial \phi} \dot{\phi}, \quad (21)$$

it can be concluded that

$$\begin{aligned} \mathbf{F}_{C_1} = -\mathbf{F}_{C_2} &= \sum_{i=1}^3 \frac{\partial U}{\partial y_i} \mathbf{p}_i, \\ \mathbf{M}_{C_1} = -\mathbf{M}_{C_2} &= \frac{\partial U}{\partial \psi} \mathbf{g}^1 + \frac{\partial U}{\partial \theta} \mathbf{g}^2 + \frac{\partial U}{\partial \phi} \mathbf{g}^3. \end{aligned} \quad (22)$$

276 These are the conservative forces and moments exerted by the disc on the vertebrae. The sim-  
 277 plicity of the representations for the conservative moments  $\mathbf{M}_{C_1}$  and  $\mathbf{M}_{C_2}$  is directly attributable  
 278 to the use of the dual Euler basis.

10. The interested reader is referred to Casey and Lam [39] where a discussion of relative angular velocity vectors can be found.

11. Their work is compatible with, and a generalization of, works on moment potentials (e.g., [40], [41]) and is entirely consistent with previous developments on moment potentials in the dynamics of rigid bodies.

12. In [24],  $X_1$  and  $X_2$  are chosen to be the centers of mass. A comparison of the expressions for the resultant moment relative to a center of mass and an arbitrary material point can be used to show that this restriction can be removed, and it is done so here without further comment.



Assuming that  $U$  is a quadratic function of  $y$  and the Euler angles, a Taylor series expansion would show that

$$U = \frac{1}{2} \begin{bmatrix} y_1 & y_2 & y_3 \end{bmatrix} \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{12} & a_{22} & a_{23} \\ a_{13} & a_{23} & a_{33} \end{bmatrix} \begin{bmatrix} y_1 \\ y_2 \\ y_3 \end{bmatrix} + \begin{bmatrix} y_1 & y_2 & y_3 \end{bmatrix} \begin{bmatrix} b_{11} & b_{12} & b_{13} \\ b_{21} & b_{22} & b_{23} \\ b_{31} & b_{32} & b_{33} \end{bmatrix} \begin{bmatrix} \psi \\ \theta \\ \phi \end{bmatrix} \\ + \frac{1}{2} \begin{bmatrix} \psi & \theta & \phi \end{bmatrix} \begin{bmatrix} c_{11} & c_{12} & c_{13} \\ c_{12} & c_{22} & c_{23} \\ c_{13} & c_{23} & c_{33} \end{bmatrix} \begin{bmatrix} \psi \\ \theta \\ \phi \end{bmatrix}. \quad (23)$$

279 This function has 21 unknown coefficients. Examples featuring the identification of these coef-  
280 ficients occupies Section 3.2 of the present paper.

281 One can view the potential energy function  $U$  as a generalization of a potential energy  
282 function for a motion segment that was proposed by Panjabi et al. [14]. Their function assumed  
283 infinitesimal rotations and was intended for use in the thoracic region of the spine. The value of  
284 present formulation is that one no longer needs to impose such kinematic restrictions. The added  
285 expense, however, is that one needs to keep track of the Euler angles during measurements of  
286 forces and moments. If one restricts attention to infinitesimal rotations, then the expressions for  
287  $\mathbf{g}^i$  simplify (cf. Eq. (16)). If one then imposes the symmetry restrictions used in [14], then the  $U$   
288 presented in Eq. (23) would reduce to the function proposed by Panjabi, Brand and White with  
289 its 12 coefficients.

To facilitate further comparison to the Panjabi, Brand and White function, one can compute, with the help of Eq. (22) and Eq. (23), the relationship between the conservative forces and conservative moments and the translational and angular displacements. These results are expressed in the compact form:

$$\mathbf{F}_C = -\mathbf{K}_U \mathbf{d}, \quad (24)$$

where the generalized force vector  $\mathbf{F}_C$ , generalized displacement vector  $\mathbf{d}$ , and stiffness matrix  $\mathbf{K}_U$  are

$$\mathbf{F}_C = \begin{bmatrix} \mathbf{F}_{C_2} \cdot \mathbf{p}_1 \\ \mathbf{F}_{C_2} \cdot \mathbf{p}_2 \\ \mathbf{F}_{C_2} \cdot \mathbf{p}_3 \\ \mathbf{M}_{C_2} \cdot \mathbf{g}_1 \\ \mathbf{M}_{C_2} \cdot \mathbf{g}_2 \\ \mathbf{M}_{C_2} \cdot \mathbf{g}_3 \end{bmatrix}, \quad \mathbf{d} = \begin{bmatrix} y_1 \\ y_2 \\ y_3 \\ \psi \\ \theta \\ \phi \end{bmatrix}, \quad \mathbf{K}_U = \begin{bmatrix} a_{11} & a_{12} & a_{13} & b_{11} & b_{12} & b_{13} \\ a_{12} & a_{22} & a_{23} & b_{21} & b_{22} & b_{23} \\ a_{13} & a_{23} & a_{33} & b_{31} & b_{32} & b_{33} \\ b_{11} & b_{21} & b_{31} & c_{11} & c_{12} & c_{13} \\ b_{12} & b_{22} & b_{32} & c_{12} & c_{22} & c_{23} \\ b_{13} & b_{23} & b_{33} & c_{13} & c_{23} & c_{33} \end{bmatrix}. \quad (25)$$

290 The corresponding forces  $\mathbf{F}_{C_1}$  and moments  $\mathbf{M}_{C_1}$  on  $\mathcal{V}_1$  are equal and opposite to  $\mathbf{F}_{C_2}$  and  $\mathbf{M}_{C_2}$ ,  
291 respectively (cf. Eq. (22)). It needs to be emphasized that the components of the moments in Eq.  
292 (24) are taken relative to the Euler basis:  $\mathbf{M}_{C_2} = -\mathbf{M}_{C_1} = \sum_{k=1}^3 (\mathbf{M}_{C_2} \cdot \mathbf{g}_k) \mathbf{g}^k$ .

## 293 B.2 Viscous Forces and Viscous Moments

It is well-known that the intervertebral disc is a viscoelastic body and consequently any model for the motion segment must accommodate this behavior. Here, the simplest possible viscous terms are considered and it is assumed that the viscoelastic forces ( $\mathbf{F}_{V_1}$  and  $\mathbf{F}_{V_2}$ ) and moments ( $\mathbf{M}_{V_1}$  and  $\mathbf{M}_{V_2}$ ) have the representations

$$\mathbf{F}_{V_2} = -\mathbf{F}_{V_1} = -c_1 \dot{y}_1 \mathbf{p}_1 - c_2 \dot{y}_2 \mathbf{p}_2 - c_3 \dot{y}_3 \mathbf{p}_3, \\ \mathbf{M}_{V_2} = -\mathbf{M}_{V_1} = -d_1 \dot{\psi} \mathbf{g}^1 - d_2 \dot{\theta} \mathbf{g}^2 - d_3 \dot{\phi} \mathbf{g}^3. \quad (26)$$

It is easy to motivate the assumption that the constants  $d_k$  and  $c_k$  are non-negative by examining the combined power  $\mathcal{P}$  of these forces and moments:<sup>13</sup>

$$\begin{aligned}\mathcal{P} &= \sum_{i=1}^2 (\mathbf{F}_{V_i} \cdot \dot{\mathbf{x}}_i + \mathbf{M}_{V_i} \cdot \boldsymbol{\omega}_i) \\ &= - \left( \sum_{k=1}^3 c_k \dot{y}_k \dot{y}_k \right) - d_1 \dot{\psi}^2 - d_2 \dot{\theta}^2 - d_3 \dot{\phi}^2.\end{aligned}\quad (27)$$

Clearly,  $\mathcal{P} \leq 0$  if  $d_k \geq 0$  and  $c_k \geq 0$ . More complex forms of the forces and moments shown in Eq. (26) are eminently possible, but these suffice for the present purposes. It is also important to note that even if the  $d_k$ 's had equal value, neither  $\mathbf{M}_{V_2}$  nor  $\mathbf{M}_{V_1}$  are necessarily parallel to  $\boldsymbol{\omega}_2 - \boldsymbol{\omega}_1$ . The viscous and conservative components of the forces and moments can be additively combined to obtain the viscoelastic forces due to the vertebral joint: e.g.,  $\mathbf{F}_1 = \mathbf{F}_{C_1} + \mathbf{F}_{V_1}$ .

### B.3 Nonconservative Contributions

In addition to the aforementioned viscoelastic contributions, the resultant forces and moments experienced by the vertebra will also include nonconservative contributions due to the contact forces in the facet joints and activation forces in the ligaments. Labelling these nonconservative contributions with the subscript  $nc$ , one has the following expressions for the resultant forces and moments:

$$\mathbf{F}_k = \mathbf{F}_{nc_k} + \mathbf{F}_{C_k} + \mathbf{F}_{V_k}, \quad \mathbf{M}_k = \mathbf{M}_{nc_k} + \mathbf{M}_{C_k} + \mathbf{M}_{V_k}, \quad (28)$$

where  $k = 1, 2$ . As with the previous developments  $\mathbf{F}_1 = -\mathbf{F}_2$  and  $\mathbf{M}_1 = -\mathbf{M}_2$ .

### B.4 The Stiffness Matrix of the Vertebral Unit

To accommodate these residual forces and moments, one performs a Taylor series expansion of the forces  $\mathbf{F}_1$  and  $\mathbf{F}_2$  and moments  $\mathbf{M}_1$  and  $\mathbf{M}_2$ . Truncating this expansion at second order, ignoring the viscous contribution, leads to a representation of the form shown in Eq. (2) for  $\mathbf{F}_2$  and  $\mathbf{M}_2$

## APPENDIX C

### TRANSFORMING MOMENTS

It is often desired to transform the components of a vector with respect to the basis  $\{\mathbf{p}_1, \mathbf{p}_2, \mathbf{p}_3\}$  to the corresponding components with respect to the bases  $\{\mathbf{g}^1, \mathbf{g}^2, \mathbf{g}^3\}$  and  $\{\mathbf{t}_1, \mathbf{t}_2, \mathbf{t}_3\}$ . Denoting this vector by  $\mathbf{b}$ , the following representations of this vector can be defined:

$$\mathbf{b} = \sum_{k=1}^3 B_k \mathbf{p}_k = \sum_{k=1}^3 b_k \mathbf{t}_k = \sum_{k=1}^3 \beta_k \mathbf{g}^k. \quad (29)$$

Then, with the help of Eqs. (1), (10), and (15), one finds that

$$\begin{aligned}\begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix} &= \begin{bmatrix} R_{11} & R_{21} & R_{31} \\ R_{12} & R_{22} & R_{32} \\ R_{13} & R_{23} & R_{33} \end{bmatrix} \begin{bmatrix} B_1 \\ B_2 \\ B_3 \end{bmatrix}, \\ \begin{bmatrix} \beta_1 \\ \beta_2 \\ \beta_3 \end{bmatrix} &= \begin{bmatrix} 0 & 0 & 1 \\ -\sin(\psi) & \cos(\psi) & 0 \\ \cos(\theta) \cos(\psi) & \cos(\theta) \sin(\psi) & -\sin(\theta) \end{bmatrix} \begin{bmatrix} B_1 \\ B_2 \\ B_3 \end{bmatrix}.\end{aligned}\quad (30)$$

In the interests of brevity, the reader is referred to Eq. (11) where expressions for the components  $R_{ik}$  can be found.

13. This calculation is facilitated by the fact that the dual Euler basis was used to establish representations for  $\mathbf{M}_{V_1}$  and  $\mathbf{M}_{V_2}$ .

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